

SUPPLEMENTAL TYPE CERTIFICATE

10087533

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

MT-PROPELLER ENTWICKLUNG GmbH

FLUGPLATZSTRASSE 1
94348 ATTING
GERMANY
EASA.21J.020

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: US A25CE

Type Certificate Holder: TEXTRON AVIATION INC.

Type: 404/406

Model: 404

Description of Design Change:

Installation of MTV-16-1-E-C-F/CF225-37() Propeller on Textron Aviation Inc. Model 404.

EASA Certification Basis:

The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:

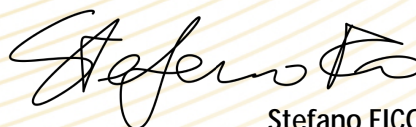
- The following paragraph(s) at a later amendment: CS-36 Amdt 6, 14 CFR Part 36 Amdt 36-32.

This certificate/approval involves a change with an impact on the noise database and/or emission levels. The environmental protection requirements are amended as follows: CS-36 Amdt 6, 14 CFR Part 36 Amdt 36-32.

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 08 July 2025



Stefano FICO
Section Manager
Business Jets



Associated Technical Documentation:

According to MT-Propeller Master Documentation List No. E-4140 Rev. 0

To be operated in accordance with MT-Propeller AFM-S Document No. E-4142 Rev. 0

To be installed i.a.w. MT-Propeller Installation Instructions Doc. No. E-4143 Rev. 0

To be maintained i.a.w. MT-Propeller ICA Doc. No. E-4144 Rev. 0

or later revisions of the above listed document(s) approved/accepted under the EASA system.

Limitations/Conditions:

According to limitation section of AFM-S.

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -