

SUPPLEMENTAL TYPE CERTIFICATE

10086373

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

MT-PROPELLER ENTWICKLUNG GmbH

FLUGPLATZSTRASSE 1
94348 ATTING
GERMANY
EASA.21J.020

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: US A1CE

Type Certificate Holder: BEECHCRAFT CORPORATION

Type: Hawker Beechcraft 19, 23, 24

Model: B24R

C24R

Description of Design Change:

Installation of MTV-12-B/188-59() Propeller on Beechcraft B24R and C24R.

EASA Certification Basis:

The Type and OSD Certification Bases for the original product as amended by the following additional or alternative airworthiness requirements:
the following paragraph(s) at a later amendment: CS-36 Amdt 6, 14 CFR Part 36 Amdt 36-32

This certificate/approval involves a change with an impact on the noise database and/or emission levels. The environmental protection requirements are amended as follows: CS-36 Amdt 6, 14 CFR Part 36 Amdt 36-32


Associated Technical Documentation:

According to MT-Propeller Master Documentation List No. E-4281 Rev. 0

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 06 February 2025



Marco CAPACCIO
Section Manager

Small Aircraft, Balloons & Airships



To be operated in accordance with MT-Propeller AFM-S Document No. E-4283 Rev. 0
To be installed i.a.w. MT-Propeller Installation Instruction Doc. No. E-4284 Rev. 0
To be maintained i.a.w. MT-Propeller ICA Doc. No. E-4285 Rev. 0
or later revisions of the above listed document(s) approved/accepted under the EASA system.

Limitations/Conditions:

According limitation section of AFM-S

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -

