

SUPPLEMENTAL TYPE CERTIFICATE

10071814 REV. 1

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

MT-PROPELLER ENTWICKLUNG GmbH

FLUGPLATZSTRASSE 1
94348 ATTING
GERMANY
EASA.21J.020

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.IM.A.239
US A350

Type Certificate Holder: Piper Aircraft, Inc.

Type: Piper PA-32

Model: PA-32-260 (Cherokee Six 260)

PA-32-300 (Cherokee Six 300)

PA-32-301 (Saratoga)

PA-32R-300/-301

PA-32RT-300

Original STC Number: EASA.A.S.00748

Description of Design Change:

Installation of MTV-9-B/198-52() on Piper PA-32

EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

the following paragraph(s) at a later amendment: CS-36, Amendment 4, FAR 36, Amdt 36-30

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 26 November 2019



Dominique ROLAND
Head of Department
General Aviation

This certificate/ approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.

Associated Technical Documentation:

According to MT-Propeller Master Documentation List No. E-3504, Rev 1

To be operated in accordance with MT-Propeller AFM-S Document No. E-3506, Rev 1

To be operated in accordance with MT-Propeller AFM-S Document No. E-3525, Rev 1 with NR Kit MTP 198-52-PA32®

To be installed i.a.w. Installation Instructions Doc. No. E-3507, Rev 1, or later revisions or later revisions of the above listed document(s) approved/accepted under the EASA system.

Limitations/Conditions:

According limitation section of AFM-S

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -

