

# SUPPLEMENTAL TYPE CERTIFICATE

### 10070901

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

## MT-PROPELLER ENTWICKLUNG GmbH

FLUGPLATZSTRASSE 1 94348 ATTING GERMANY

EASA.21J.020

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: US A28CE

Type Certificate Holder: TEXTRON AVIATION INC.

Type: 441 Model: 441

#### **Description of Design Change:**

Installation of MTV-27-1-E-C-F-R(G)/CFR() on Cessna 441

#### **EASA Certification Basis:**

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

### **Associated Technical Documentation:**

- According to MT-Propeller Master Documentation List No. E-3416 Rev. 0
- To be operated in accordance with MT-Propeller AFM-S Document No. E-3418 Rev. 0
- To be installed i.a.w. Installation Instruction Doc. No. E-3419 Rev. 0
- To be maintained i.a.w. MT-Propeller ICA Document No. E-3420 Rev. 0

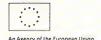
or later revisions of the above listed document(s) approved/accepted under the EASA system.

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 05 September 2019

Dominique ROLAND
Head of Department
General Aviation







### **Limitations/Conditions:**

According limitation section of AFM-S.

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -