

SUPPLEMENTAL TYPE CERTIFICATE

10064631 REV. 1

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

MT-PROPELLER ENTWICKLUNG GmbH

FLUGPLATZSTRASSE 1
94348 ATTING
GERMANY

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.IM.A.279

US 3A15

Type Certificate Holder: TEXTRON AVIATION INC.

Type: 33, 35, 36 (Bonanza)

Model: 35-C33A, 36, A36, E33A, E33C

F33A

F33C, S35, V35 (TC), V35A (TC)

V35B (TC), G36, A36TC, B36TC

Description of Design Change:

Installation of 3-blade constant speed propeller MTV-9-D/200-58 on Textron Aviation Inc. Bonanza Models

EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

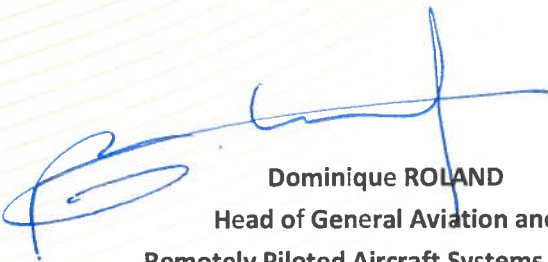
CS-36, Amendment 4

FAR 36, Amendment 36-30

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 16 February 2018



Dominique ROLAND
Head of General Aviation and
Remotely Piloted Aircraft Systems (RPAS)



This certificate/ approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.

Associated Technical Documentation:

According to MT-Propeller Master Documentation List No. E-2934, Rev 0

To be operated in accordance with MT-Propeller AFM-S Document No. E-2931, Rev 0
or later revisions of the above listed documents approved by EASA.

To be installed i.a.w. Installation Instructions Doc. No. E-2932, Rev 0, or later revisions

To be maintained i.a.w. Instruction for continued airworthiness Doc. No. E-2933, Rev 0, or later revisions

Limitations/Conditions:

According limitation section of AFM-S

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -

