

## SUPPLEMENTAL TYPE CERTIFICATE

**10063415**

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

### **MT-PROPELLER ENTWICKLUNG GmbH**

**FLUGPLATZSTRASSE 1  
94348 ATTING  
GERMANY**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

**Original Type Certificate Number:** EASA.A.576

**Type Certificate Holder:** COSTRUZIONI AERO. TECNAM

**Type:** P2010

**Model:** P2010

**Description of Design Change:**

Installation of MTV-12() propeller on Tecnam P2010.

**EASA Certification Basis:**

Installation of MTV-12() propeller on Tecnam P2010

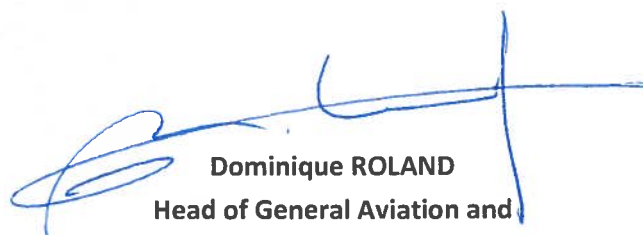
The following paragraph(s) at a later amendment: CS-36, Amendment 4, FAR 36, Amendment 36-30.

This certificate/ approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.

See Continuation Sheet(s)

**For the European Aviation Safety Agency**

**Cologne, Germany, 12 October 2017**



**Dominique ROLAND**  
Head of General Aviation and  
Remotely Piloted Aircraft Systems (RPAS)

**Associated Technical Documentation:**

According to MT-Propeller Master Documentation List No. E-2982, Rev 0,  
To be operated in accordance with MT-Propeller AFM-S Document No. E-2979, Rev.0,

or later revisions of the above listed documents approved by EASA.

To be installed i.a.w. Installation Instruction Doc. No. E-2980, Rev 0, or later Revisions.  
To be maintained i.a.w. Instruction for continued airworthiness and 2981, Rev 0, or later Revisions.

**Limitations/Conditions:**

According AFM-S.

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -