

# SUPPLEMENTAL TYPE CERTIFICATE

### 10060053

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

## **MT-PROPELLER ENTWICKLUNG GmbH**

FLUGPLATZSTRASSE 1 94348 ATTING GERMANY

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: US 3A21 Type Certificate Holder: TEXTRON AVIATION INC. Type: 210 SERIES Model: P210N

Description of Design Change:

MTV-5-1-D-C-F-R(A)/CFR210-56 propeller on Textron P210N modified as per LBA STC 0779/625b (FAA STC SA1003NE).

EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

The following paragraph(s) at a later amendment: CS-36, Amendment 3 - FAR 36, Amendment 36-30.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Date of Issue: 11 November 2016

Dominique ROLAND Head of General Aviation and

Remotely Piloted Aircraft Systems (RPAS)

10042100

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This certificate/ approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.

### Associated Technical Documentation:

- According to MT-Propeller Master Documentation List No. E-2908, Rev 0

- To be operated in accordance with MT-Propeller AFM-S Document No. E-2905, Rev 0 or later revisions of the above listed documents approved by EASA.

- To be installed i.a.w. Installation Instruction Doc. No. E-2906, Rev 0, or later Revisions.

- To be maintained i.a.w. MT-Propeller ICA Document No. E-2907, Rev 0, or later Revisions.

#### Limitations/Conditions:

According limitation section of AFM-S. Prerequisite: LBA STC 0779/625b or FAA STC SA1003NE

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -

An Agency of the Europe an Union