

## SUPPLEMENTAL TYPE CERTIFICATE

**10054491 REV. 1**

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

### **MT-PROPELLER ENTWICKLUNG GmbH**

FLUGPLATZSTRASSE 1  
94348 ATTING  
GERMANY  
EASA.21J.020

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

**Type Certificate Number:** FAA A1250

**Type Certificate Holder:** COMMANDER AIRCRAFT CORPORATION

**Type:** COMMANDER 112/114 SERIES

**Model:** 112

112B

112TC

112TCA

**Description of Design Change:**

Installation of 3-blade constant speed propeller MTV-12-B/188-59f on Commander Aircraft Corp. 112 aircraft models

**EASA Certification Basis:**

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

CS 36, Amdt 3 and FAR 36 Amdt 30

See Continuation Sheet(s)

**For the European Aviation Safety Agency**

**Cologne, Germany, 03 May 2018**



PP

**Dominique ROLAND**  
Head of General Aviation and  
Remotely Piloted Aircraft Systems (RPAS)



This certificate/ approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.

**Associated Technical Documentation:**

According to MT-Propeller Master Document List No. E-3306

To be operated in accordance with MT-Propeller AFM-S Document No. E-2612

To be installed iaw. Installation Instructions Doc. No. E-2613

To be maintained i.a.w. MT-Propeller ICA Document No. E-2614

or later revisions of the above listed document(s) approved under the EASA system.

**Limitations/Conditions:**

According limitation section of AFM-S

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -