

SUPPLEMENTAL TYPE CERTIFICATE

10054487 REV. 1

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

MT-PROPELLER ENTWICKLUNG GmbH

FLUGPLATZSTRASSE 1 94348 ATTING GERMANY

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: US A4CE

Type Certificate Holder: Textron Aviation Inc.

Type: 206 SERIES

Model: P206, P206A, P206B, P206C

P206D, P206E, U206, U206A

TP206A, TP206B, TP206C

TP206D, TP206E, TU206A

TU206B, TU206C, TU206D

TU206E, TU206F, TU206G

U206B, U206C, U206D, U206E

U206F, U206G, 206

Description of Design Change:

Installation of 3-blade constant speed propeller MTV-9-D/200-58 on Textron Aviation Inc. Models 206, U206, U206A through G, P206, P206A through E, TU206A through G and TP206A through E.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 06 February 2018

Dominique ROLAND
Head of General Aviation and

Remotely Piloted Aircraft Systems (RPAS)

10035564

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EASA Certification Basis:

The Certification Basis for the original product as amended by the following additional or alternative airworthiness requirements:

The following paragraph(s) at a later amendment:

CS-36, Amendment 4.

FAR 36, Amendment 36-30.

This certificate/ approval involves a change to the requirements for environmental protection or a change to the certified noise or emissions levels.

Associated Technical Documentation:

According to MT-Propeller Master Documentation List No. E-2625, Rev 2; To be operated in accordance with MT-Propeller AFM-S Document No. E-2621, Rev 1;

or later revisions of the above listed documents approved by EASA.

To be installed i.a.w. Installation Instructions Doc. No. E-2622, Rev 1, or later revisions. To be maintained i.a.w. Instruction for continued airworthiness Doc. No. E-2623, Rev 1, or later revisions.

Limitations/Conditions:

According limitation section of AFM-S.

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.



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