



## SUPPLEMENTAL TYPE CERTIFICATE

**10014238, REV. 2**

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EC) No. 1702/2003 to

### **MT-PROPELLER ENTWICKLUNG GMBH**

**FLUGPLATZSTRASSE 1  
94348 ATTING  
GERMANY**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

**Original Product TC Number : 1.EASA.A.439 & 2.EASA.IM.A.223  
TC Holder : 1.DIAMOND AIRCRAFT IND GMBH  
TC Holder : 2.DIAMOND AIRCRAFT IND INC  
Model : 1. DV20 & 2. DA20-A1**

**Description of Design Change:**

Extension of EASA STC EASA.A.S.01402  
Installation of the 2-blade MTV-21-A/170-05

**EASA Certification Basis:**

The Certification Basis for the original product and the following additional or alternative airworthiness requirements are applicable to this certificate/ approval.

CS-VLA, Amendment 1 (not required by 21A.101, elected to comply by the applicant)

This certificate/ approval involves a change to the requirements for environmental protection and/ or change to the certificated noise and/ or emissions levels.

See Continuation Sheet(s)

**For the European Aviation Safety Agency,**

**Date of issue: 04.11.2010**

**Roger HARDY  
Certification Manager  
General Aviation**

Note:  
The following numbers are listed on the certificate:  
EASA old Project Number: EASA.A.S.01402, REV. 2

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**Associated Technical Documentation:**

- Aircraft Flight Manual Supplement, Doc. No. E-1909, rev 0, 14.09.2010
  - Instructions for Continued Airworthiness, Doc. No. E-1911, rev. 0, 14.09.2010
- or later revisions of the above listed documents approved by EASA

**Limitations:**

none

**Conditions:**

Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- end -

Note:  
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EASA old Project Number: EASA.A.S.01402, REV. 2

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