

## SUPPLEMENTAL TYPE CERTIFICATE

**10086020**

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

### MT-PROPELLER ENTWICKLUNG GmbH

FLUGPLATZSTRASSE 1  
94348 ATTING  
GERMANY  
EASA.21J.020

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

**Type Certificate Number:** US A17SW

**Type Certificate Holder:** AIR TRACTOR, INC.

**Type:** AT-401, -402, -500 Series

**Model:** AT-402, AT-402A, AT-402B

AT-502, AT-502B, AT-503A

#### Description of Design Change:

Installation of 3-blade Propeller V508H/108/A on Air Tractor Inc. AT-402, -402A, -402B, -502, -502B, -503A

#### EASA Certification Basis:

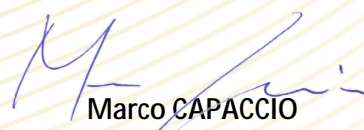
The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/approval.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact on the noise database.

See Continuation Sheet(s)

**For the European Union Aviation Safety Agency**

**Cologne, Germany, 09 December 2024**



**Marco CAPACCIO**  
Section Manager

**Small Aircraft, Balloons & Airships**





**Associated Technical Documentation:**

According to MT-Propeller Master Documentation List No. E-4385 Rev. 1

To be operated in accordance with MT-Propeller AFM-S Document No. E-4387 Rev. 1

To be installed i.a.w. MT-Propeller Installation Instruction Doc. No. E-4388 Rev. 0

To be maintained i.a.w. MT-Propeller ICA Doc. No. E-4389 Rev. 1

or later revisions of the above listed document(s) approved/accepted under the EASA system.

**Limitations/Conditions:**

According to limitation section of AFM-S.

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -